IN THE CLAIMS:

- 1-17. (canceled)
- (currently amended) A gas turbine engine assembly comprising:

a gas turbine engine comprising a compressor;

- a pump; and
- a ring manifold coupled in fluid communication with said pump, said ring manifold mounted within said gas turbine engine upstream from said compressor, said ring manifold comprising a plurality of circumferentially-spaced spray nozzles, at least one of said plurality of circumferentially-spaced spray nozzles operable to discharge a first liquid to facilitate removing particulate matter and a second liquid to facilitate reducing a rate of formation of particulate matter, wherein at least one of said plurality of circumferentially-spaced spray nozzles is oriented to discharge at least one of the first liquid and the second liquid radially inwardly such that at least a portion of said compressor is coated with the first liquid and the second liquid sharred from said spray nozzles.
- 19. (currently amended) A gas turbine engine <u>assembly</u> in accordance with Claim 18, wherein said gas turbine engine <u>assembly</u> further comprises a starter motor configured to rotate said gas turbine engine while-at-least one of the <u>first liquid and the second</u> liquid is discharged from at least one of said spray nozzles.
- (currently amended) A washing system for a gas turbine engine having a compressor, said washing system comprising:

a pump; and

a ring manifold configured to be coupled in fluid communication with said pump, said ring manifold mountable within said gas turbine engine upstream from the compressor, said ring manifold comprising a plurality of circumferentially-spaced spray nozzles, wherein at least one of said plurality of circumferentially-spaced spray nozzles is configured to be oriented to discharge liquid radially inwardly;

wherein at least one of said plurality of spray nozzles is operable to inject a first liquid to facilitate removing particulate matter and a second liquid to facilitate reducing a rate of formation of particulate matter.

- 21. (currently amended) A washing system for a gas turbine engine in accordance with Claim 20, wherein said plurality of <u>circumferentially-spaced</u> spray nozzles is configured to inject the <u>a</u> first liquid into the gas turbine engine before injecting the <u>a</u> second liquid into the gas turbine engine.
- 22. (currently amended) A washing system for a gas turbine engine in accordance with Claim 21, wherein said plurality of <u>circumferentially-spaced spray</u> nozzles is configured to inject the second liquid into the gas turbine engine such that the second liquid coats at least a portion of the gas turbine engine.
- 23. (currently amended) A washing system for a gas turbine engine in accordance with Claim 20, wherein said washing system comprises a starter motor configured to rotate the gas turbine engine while the first liquid is being discharged.